

$$1) - \begin{cases} \vec{\Omega}(R_p | R_0) = \Omega \vec{e}_3 \\ \vec{\Omega}(R_D | R_p) = \omega \vec{e}_3 \end{cases} \Rightarrow \vec{\Omega}(R_D | R_0) = \vec{\Omega}(R_D | R_p) + \vec{\Omega}(R_p | R_0) = (\Omega + \omega) \vec{e}_3$$

$$2) \vec{v}(M | R_p) = \left. \frac{d\vec{OM}}{dt} \right]_{R_p} = \left. \frac{d(\vec{OC} + \vec{CM})}{dt} \right]_{R_p} = \left. \frac{d(L\vec{e}_x + r\vec{e}_1)}{dt} \right]_{R_p} = \left. r \frac{d\vec{e}_1}{dt} \right]_{R_p} = r\omega \vec{e}_2$$

$$\vec{v}_e(M, R_p | R_0) = \vec{v}(O \in R_p | R_0) + \vec{\Omega}(R_p | R_0) \times \vec{OM} = \Omega \vec{e}_3 \times (L\vec{e}_x + r\vec{e}_1) = -\Omega(L\vec{e}_y + r\vec{e}_2)$$

$\parallel$   
 $\vec{0}$

$$\vec{v}(M | R_0) = \left. \frac{d\vec{OM}}{dt} \right]_{R_0} = L \left. \frac{d\vec{e}_x}{dt} \right]_{R_0} + r \left. \frac{d\vec{e}_1}{dt} \right]_{R_0} = L\Omega \vec{e}_y + r(\Omega + \omega) \vec{e}_2$$

$$\Rightarrow \vec{v}(M | R_0) = \vec{v}(M | R_p) + \vec{v}_e(M, R_p | R_0) \quad (\text{CQFD})$$

$$3) \vec{a}(M | R_p) = \left. \frac{d\vec{v}(M | R_p)}{dt} \right]_{R_p} = r\omega \left. \frac{d\vec{e}_2}{dt} \right]_{R_p} = -R\omega^2 \vec{e}_2$$

$$\vec{a}_c(M, R_p | R_0) = 2 \vec{\Omega}(R_p | R_0) \times \vec{v}(M | R_p) = 2\Omega \vec{e}_3 \times r\omega \vec{e}_2 = -2r\Omega\omega \vec{e}_1$$

$$\vec{a}_e(M, R_p | R_0) = \vec{a}(O_p | R_0) + \vec{\Omega}(R_p | R_0) \times \left[ \vec{\Omega}(R_p | R_0) \times \vec{O_p M} \right] + \left. \frac{d\vec{\Omega}(R_p | R_0)}{dt} \right]_{R_0} \times \vec{O_p M}$$

$$= \vec{0} + \Omega \vec{e}_3 \times \left[ \Omega \vec{e}_3 \times (L\vec{e}_x + r\vec{e}_1) \right]$$

$$= \Omega \vec{e}_3 \times (-\Omega L \vec{e}_y + \Omega r \vec{e}_2) = -\Omega^2 (L\vec{e}_x + r\vec{e}_1)$$

$$\vec{a}(M | R_0) = \left. \frac{d\vec{v}(M | R_0)}{dt} \right]_{R_0} = \left. \frac{d[L\Omega \vec{e}_y + r(\Omega + \omega) \vec{e}_2]}{dt} \right]_{R_0} = -L\Omega^2 \vec{e}_x - r(\Omega + \omega)^2 \vec{e}_1$$

$$\vec{a}(M | R_p) + \vec{a}_c(M, R_p | R_0) + \vec{a}_e(M, R_p | R_0) = \underbrace{(-R\omega^2 - 2r\Omega\omega - r\Omega^2)}_{-r(\Omega + \omega)^2} \vec{e}_1 - L\Omega^2 \vec{e}_x$$

$$= \vec{a}(M | R_0) \quad (\text{CQFD})$$

1.  $R_0 = (T, \vec{e}_{x_0}, \vec{e}_{y_0}, \vec{e}_{z_0})$  axes fixes (3 étalles) / képiac on Copernic.

2. Th. de Gauss =  $\oint_S \vec{g} \cdot \vec{n} dS = -4\pi GM_{int} \ominus$  et  $\vec{g}(r) = -g(r) \vec{e}_r$   
⑤ → sphère de centre T et rayon r

$$\Rightarrow -g(r) \cdot 4\pi r^2 = -4\pi GM \Leftrightarrow g(r) = + \frac{GM}{r^2} = + \frac{GM}{(R+z)^2}$$
$$g_0 = g(r) = + \frac{GM}{R^2} \Rightarrow GM = g_0 R^2 \left. \vphantom{g_0} \right\} \rightarrow g(r) = g_0 \frac{R^2}{(R+z)^2}$$

3.  $\Sigma \vec{F} = m\vec{a} = -m \frac{v^2}{r} \vec{e}_r + m \frac{dv}{dt} \vec{e}_\theta$

$$\Leftrightarrow - \frac{GMm}{r^2} \vec{e}_r = - \frac{mv^2}{r} \vec{e}_r \quad \text{et} \quad \frac{dv}{dt} = 0 \Rightarrow V = \sqrt{\frac{GM}{r}} = \sqrt{\frac{g_0 R^2}{r}}$$

4.  $T = \frac{2\pi r}{v} = \frac{2\pi}{R} \sqrt{\frac{r^3}{g_0}}$

5.  $E_k = \frac{1}{2} m v^2 = \frac{1}{2} \frac{GMm}{r} = \frac{1}{2} m g_0 \frac{R^2}{r}$ ,  $E_p = - \frac{GMm}{r} = - m g_0 \frac{R^2}{r} = -2E_k$

6.  $E_m = E_k + E_p = - \frac{1}{2} m g_0 \frac{R^2}{r} < 0 \Rightarrow$  état lié.

7. Géostationnaire s'il est tjs à la verticale du m pt de la Terre.

8. Mut à force centrale  $\Rightarrow$  trajectoire ds un plan  $\ni$  le centre de force, et  $\perp \vec{\omega}$

$\Rightarrow$  mut ds le plan équatorial. (géost  $\Rightarrow T=24h \Rightarrow h=35.930 \text{ km}$ )

9. NON - Paris  $\notin$  plan équatorial.